Proportional Derivative (PD) Control on the Euclidean Group*

Francesco Bullo†
bullo@indra.caltech.edu
Richard M. Murray
murray@indra.caltech.edu
Division of Engineering and Applied Science
California Institute of Technology
Pasadena, CA 91125

Keywords: PD control, Euclidean group, nonlinear control, workspace control

ABSTRACT

In this paper we study the stabilization problem for control systems defined on $SE(3)$, the Euclidean group of rigid-body motions. Assuming one actuator is available for each degree of freedom, we exploit geometric properties of Lie groups (and corresponding Lie algebras) to generalize the classical PD control in a coordinate-free way. For the $SO(3)$ case, the compactness of the group gives rise to a natural metric structure and to a natural choice of preferred control direction: an optimal (in the sense of geodesic) solution is given to the attitude control problem. In the $SE(3)$ case, no natural metric is uniquely defined, so that more freedom is left in the control design. Different formulations of PD feedback can be adopted by extending the $SO(3)$ approach to the whole of $SE(3)$ or by breaking the problem into a control problem on $SO(3) \times \mathbb{R}^3$. For the simple $SE(2)$ case, simulations are reported to illustrate the behavior of the different choices. Finally, we discuss the trajectory tracking problem and show how to reduce it to a stabilization problem, mimicking the usual approach in $\mathbb{R}^n$.

I. INTRODUCTION

We here consider the problem of controlling a (mechanical) system whose configuration space is a matrix Lie group: we focus on second order systems and attempt to generalize the standard notion of proportional derivative feedback. One large class of applications which motivates this work is workspace control of robotic manipulators, where the end-effector configuration is naturally embedded in $SE(3)$ (see [1] for a description of the workspace control problem and traditional solutions). While local solutions are easily obtained, we hope that a more geometric approach will yield advantages similar to those afforded by the geometric approach to kinematics in [1].

Historically, nonlinear control systems defined on Lie groups have received considerable attention in the literature: early work by Brockett [2, 3], and others has served as motivation for more recent contributions by Walsh, Sarti, Sastry and Montgomery [4, 5], Leonard and Krishnaprasad [6], and Crouch and Silva Leite [7], to name a few. Early works concentrated on problem formulation and controllability issues, while the more recent papers mainly consider constructive controllability: how to generate a feasible trajectory between two (or more) points on the configuration manifold given a limited number of actuators.

Our approach in this paper is somewhat different. We concentrate on the problems of stabilization and trajectory tracking in the fully actuated case, where one actuator is available for each degree of freedom in the system. This is traditionally the situation for problems in robotic manipulation, satellite reorientation and 6 degree of freedom underwater vehicles. We attempt to exploit the geometric properties of Lie groups and to generalize the classical proportional plus derivative feedback (PD) used for control of simple mechanical systems in $\mathbb{R}^n$. For the case of compact Lie groups, such as $SO(3)$, our results are completely general. For the non-compact case, we consider only control systems on $SE(3)$ and on its subgroups, since those are the main systems of interest in our applications.

The paper is organized as follows. In Section II, we introduce basic and new results on systems defined on Lie groups. Section III shows stabilization results for the compact case and in particular for $SO(3)$. Section IV considers the $SE(3)$ case, a non-compact non-semisimple group. Different metrics lead to different control laws. These results are then generalized to the trajectory tracking case in Section V. Section VI discusses the results and the underactuated case. For all the proofs and for a more detailed version of this paper we refer to [8].

II. SYSTEMS ON LIE GROUPS

We here review the notations and give some algebraic results on Lie groups and on dynamical systems evolving on Lie groups. For a comprehensive introduction see [1].

II.1. Basic definitions and results In the following we focus our attention on the matrix Lie group $SE(3)$ and its proper subgroups, even though most of the results hold more generally. Let $G \subset SE(3)$ be a matrix Lie group and $g \subset \mathfrak{se}(3)$ its Lie algebra. A dynamical system with state $g \in G$ evolves following

$$\dot{g} = gV^b = V^s g, \quad V^b, V^s \in \mathfrak{g},$$

where we can express the velocity in body ($V^b$) or in spatial frame ($V^s$). Since the system $\dot{g} = gV^b$ is invariant under left multiplications by constant matrices, we call it left invariant; correspondingly $\dot{g} = V^s g$ is said to be right
invariant. For all $g \in G$ and all $X, Y \in \mathfrak{g}$, the adjoint map $\text{Ad}_g$ and the matrix commutator $\text{ad}_X$ are defined as

$$\text{Ad}_g(Y) = gYg^{-1},$$

$$\text{ad}_X(Y) = [X, Y] = XY - YX.$$

On $\text{SE}(3)$ and $\text{so}(3)$ we represent a group element $g = (R, p) \in \text{SO}(3) \times \mathbb{R}^3$ and a vector $V = (\omega, v) \in \text{so}(3) \times \mathbb{R}^3$ using homogeneous coordinates,

$$g = \begin{bmatrix} R & p \\ 0 & 1 \end{bmatrix}, \quad \text{and} \quad V = \begin{bmatrix} \omega \\ v \end{bmatrix},$$

where the operator $\tilde{\times} : \mathbb{R}^3 \rightarrow \mathfrak{so}(3)$ is defined so that $\tilde{x}y = x \times y$ for all $x, y \in \mathbb{R}^3$.

On $\text{SE}(3)$ and on its proper subgroups the exponential map $\exp : \mathfrak{g} \rightarrow G$ is a surjective map and a local diffeomorphism. Standard computations show:

**Lemma 1.** (Exponential Map) Given $\tilde{\psi} \in \mathfrak{so}(3)$ and $X = (\psi, p) \in \mathfrak{se}(3)$,

$$\exp_{\mathfrak{so}(3)}(\tilde{\psi}) = I + \frac{\sin \| \psi \|}{\| \psi \|} \tilde{\psi} + \frac{1 - \cos \| \psi \|}{\| \psi \|^2} \tilde{\psi}^2,$$

$$\exp_{\mathfrak{se}(3)}(X) = \begin{bmatrix} \exp_{\mathfrak{so}(3)}(\tilde{\psi}) & A(\psi)p \\ 0 & 1 \end{bmatrix},$$

where $\| \psi \|$ is the standard Euclidean norm and

$$A(\psi) = I + \frac{1 - \cos \| \psi \|}{\| \psi \|^2} \tilde{\psi} + \frac{\| \psi \| - \sin \| \psi \|}{\| \psi \|^3} \tilde{\psi}^2.$$

In an open neighborhood of the origin dense in $G$, we define $X = \log(g) \in \mathfrak{g}$ to be the exponential coordinates of the group element $g$ and we regard the logarithmic map as a local chart of the manifold $G$.

**Lemma 2.** (Logarithmic map) Given $(R, p) \in \text{SO}(3) \times \mathbb{R}^3$ such that $\text{tr}(R) \neq -1$. Then

$$\log_{\text{SO}(3)}(R) = \frac{\phi}{2 \sin \phi} (R - R^T) \in \mathfrak{so}(3),$$

where $\phi$ satisfies $\cos \phi = (\text{tr}(R) - 1)/2$ and $|\phi| < \pi$. Also

$$\log_{\text{SE}(3)}(R, p) = \begin{bmatrix} \tilde{\psi} & A^{-1}(\psi)p \\ 0 & 1 \end{bmatrix} \in \mathfrak{se}(3),$$

where $\tilde{\psi} = \log_{\text{SO}(3)}(R)$ and

$$A^{-1}(\psi) = I - \frac{1}{2} \tilde{\psi} + \frac{2}{\| \psi \|^2} \tilde{\psi} - \| \psi \| (1 + \cos \| \psi \|) \tilde{\psi}^2.$$

Note that elements of the Lie algebra $\mathfrak{g}$ can represent a velocity as in equation (1) or can represent the matrix logarithm of the state (and should therefore be considered states) as in equation (2). We denote them with $V = (\omega, v)$ in the first case and with $X = (\tilde{\psi}, p)$ in the second.

**II.2. The Jacobian of the exponential map** We now want to compute explicit formulas that relate the time derivative of $X(t) = \log(g(t))$ with the body and spatial velocities $V^b, V^s$. Indeed only for the linear time dependence case $(X(t) = Yt)$, it is easy to show that $\dot{X} = Y = V^b = V^s$; for the generic case $X = X(t)$ the relationship is not trivial.

**Theorem 3.** (Differential of Exponential) Let $g(t)$ be a smooth curve on $G$, $X(t) = \log(g(t))$ be the exponential coordinates of $g(t)$, $V^b = g^{-1} \dot{g}$ the body velocity and $V^s = \dot{g}g^{-1}$ the spatial velocity. Then it holds

$$\dot{X} = \sum_{n=0}^{\infty} \frac{(-1)^n B_n}{n!} \text{ad}^n_X(V^b)$$

$$= \sum_{n=0}^{\infty} \frac{B_n}{n!} \text{ad}^n_X(V^s),$$

where $\{B_n\}$ are the Bernoulli numbers.

In the pure rotation case, summing the series we obtain:

**Lemma 4.** (Exponential coordinates on $\text{SO}(3)$) Let $R(t)$ be a smooth curve on $\text{SO}(3)$, $\tilde{x}(t) = \log(R(t))$ be the exponential coordinates of $R(t)$ and $\tilde{\omega} = R^{-1} \dot{R}$ the body angular velocity. Then we have

$$\dot{x} = \omega \| x \| \omega + \frac{1}{2} (\omega \times x)$$

where $\beta(y) = \frac{y^2}{\tan(y/2)}$ and $\omega = \omega_\| + \omega_\perp$ is the orthogonal decomposition of $\omega$ along span $\{x\}$ and span $\{x\}^\perp$.

To the authors’ knowledge this expression is novel and relates the time derivative of the angle-axis quantity $x$ with the often used $\omega$. Similar expressions can be obtained for the $\text{SE}(3)$ and $\text{SE}(2)$ cases (for more details see [8]).

**II.3. Metric properties on compact Lie groups** On any Lie group $G$, the Killing form $\langle \cdot, \cdot \rangle_K$ is defined as the bilinear operator on $\mathfrak{g} \times \mathfrak{g}$:

$$\langle X, Y \rangle_K \triangleq \text{tr}(\text{ad}_X \cdot \text{ad}_Y) \quad \forall X, Y \in \mathfrak{g}.$$

A Lie group is said to be semi-simple if $\langle \cdot, \cdot \rangle_K$ is nondegenerate. For compact Lie groups $\langle \cdot, \cdot \rangle_K$ is both nondegenerate and negative definite, so that by a simple multiplication with a negative constant, we can define an inner product on the Lie algebra $\mathfrak{g}$ (e.g. on $\text{so}(3)$ $\langle \cdot, \cdot \rangle \triangleq -1/4 \langle \cdot, \cdot \rangle_K$). An inner product defined this way will satisfy the crucial property of $G$-invariance:

$$\langle X, Y \rangle = \langle \text{Ad}_g X, \text{Ad}_g Y \rangle, \quad \forall g \in G,$$

where $\text{Ad}$ is therefore an orthogonal operator of $\mathfrak{g}$. Equivalently the matrix commutator satisfies

$$\langle \text{ad}_Z X, Y \rangle = -\langle X, \text{ad}_Z Y \rangle \quad \forall Z \in \mathfrak{g}. \tag{2}$$

Now, an $G$-invariant inner product on the algebra $\mathfrak{g}$ induces an $G$-invariant metric on the group $G$ by either left or right translation; this gives the additional structure of a Riemannian manifold to the group $G$. Without entering details, we refer to [9] and we simply state the following result:

**Proposition 5.** With respect to an $G$-invariant metric, the geodesics of $G$ are the one parameter subgroups, that is the curves of the form $\exp(Yt)$, with $Y \in \mathfrak{g}$ constant. Furthermore, the distance between the element $g$ and the identity $e_G = I \in G$ is given by the norm of the logarithmic function:

$$\|g\|_G = \langle \log(g), \log(g) \rangle^{1/2}. \tag{3}$$
The computational result we are interested in, is an extension of Gauss’s Lemma (see [9] and [10]), obtained thanks to property (2) and equation (3).

**Theorem 6.** (Derivative of distance function) Let $G$ be a compact Lie group with bi-invariant metric $(\cdot, \cdot)$. Consider a smooth trajectory $g(t) \in G$, such that $g(t)$ never passes through a singularity of the exponential map. Then

$$\frac{1}{2} \frac{d}{dt} \|g\|^2_G = \langle \log(g), V^b \rangle = \langle \log(g), V^a \rangle.$$

### III. PD CONTROL ON $SO(3)$

We begin with the problem of stabilizing a control system evolving on a compact, semi-simple Lie group. Without loss of generality we will here consider only the $SO(3)$ case. As explained in the previous section, a bi-invariant Riemannian metric is naturally defined on $SO(3)$ and allows us to easily design appropriate Lyapunov functions.

We begin by briefly describing our approach for a simple first order system on $SO(3)$, described as in equation (1) by $g = gV^b$. Consider the natural candidate Lyapunov function

$$W(g) = \frac{1}{2} \|g\|^2_{SO(3)},$$

and assume we can directly control the quantity $V^b \in so(3)$ to any desired value (i.e. the system is fully actuated). Then the proportional control action

$$V^b = -k_p \log(g), \quad k_p > 0,$$

leads to

$$W(g(t)) = \langle \log(g), -k_p \log(g) \rangle = -2k_p W,$$

thanks to Theorem 6. Thus, for this first order system, a logarithmic control law ensures exponential stability for all initial conditions $g(0)$ such that $\text{tr}(g(0)) \neq -1$.

Now, motivated by standard control problems in mechanics and robotics, we consider the controllability problem for second order systems, that is for systems where we have full control over forces (accelerations) rather than velocities. A second order system on $SO(3)$ has the form

$$\begin{cases}
\dot{g} = gV^b \\
\dot{V}^b = f(g, V^b) + U,
\end{cases}$$

where $g \in SO(3)$ is the configuration of the system, $f(g, V^b) \in so(3)$ is the internal drift, and $U \in so(3)$ is the control input. Note that we once again assume that the system is fully actuated. To regulate the state $g$ to the identity matrix $I \in SO(3)$, we couple the proportional action (4) with a derivative term, i.e. with a term proportional to the velocity $V^b$.

**Theorem 7.** (PD Control on $SO(3)$) Consider the system in equation (5) and let $K_p$ and $K_d$ be symmetric, positive definite gains. Then the control law

$$U = -f(g, V^b) - K_p \log(g) - K_d V^b,$$

exponentially stabilizes the state $g$ at $I \in SO(3)$ from any initial condition $\text{tr}(g(0)) \neq -1$ and for all $K_p$ and $V^b(0)$ such that

$$\lambda_{\min}(K_p) > \frac{\|V^b(0)\|^2}{\pi^2 - \|g(0)\|^2_{SO(3)}}$$

where $\lambda_{\min}(K_p)$ is the minimum eigenvalue of $K_p$.

**Proof.** The stability analysis is based on the candidate Lyapunov function

$$W = \frac{1}{2} \left[ \log(g) \right] V^b + \left[ \frac{\|g\|^2}{\pi^2 - \|g(0)\|^2_{SO(3)}} \right] \left[ \log(g) \right] V^b,$$

where the inner product is taken in $so(3) \times so(3)$ and $\epsilon$ is taken small enough. For the details we refer to [8].

**Remark 8.** We have written the control law (4) and Theorem 7 in terms of the body velocity $V^b$, i.e. we assumed “body-fixed” control inputs. A dual version can be easily written for the opposite case of “spatial-fixed” control inputs, i.e. for the case $V^a = f(g, V^a) + U$. Thanks to Theorem 6 a logarithmic control law is the correct choice also for this case.

#### III.1. Example: orientation control of a satellite

The primary example of control problem on a compact Lie group is attitude control of a satellite.

In the literature, various PD control laws based on different parametrization of the manifold $SO(3)$ have been proposed: Euler angles [11], Gibb’s vectors [12] and unit quaternions [13]. In particular, Wen and Kreutz-Delgado [13] introduce the idea that the “error measure should correspond to the topology of the error space”. Here we additionally require that the error measure correspond to the (natural) metric of the Riemannian manifold $SO(3)$.

The second order model of a satellite is

$$\begin{cases}
\dot{g} = g \dot{w}^b \\
J \dot{\omega}^b = f(g, \omega^b) + \tau,
\end{cases}$$

where the control inputs $\tau$ is the total torque applied to the satellite either by momentum wheels or by gas jet actuators. The internal drift is

$$f(g, \omega^b) = \begin{cases}
[g \tau m_0, \omega^b] & \text{momentum wheels} \\
[J \omega^b, \omega^b] & \text{gas jet (Euler equations)}.
\end{cases}$$

Following early work by Koditschek [14], we introduce a slight modification to the design of Theorem 7 and we adopt the modified Lyapunov function

$$W = \frac{1}{2} \|g\|^2_{SO(3)} + \frac{1}{2} \|\omega^b, J \omega^b \|_w + \epsilon \langle \log(g), J \omega^b \rangle$$

where the second term has the strong interpretation of kinetic energy. This leads to the feedback law

$$\tau = -k_p \log(g) - K \omega^b,$$

where we write the control law in $\mathbb{R}^3$ making use of the isomorphism $\sim$ given in Section II.

This feedback has strong similarities to the ones already proposed in the literature: it is instructive to compare it with the equivalent proposed in [13]. Both laws consist of the sum of a proportional and derivative action, where they differ is in the expression of the proportional term. In particular along the “geodesic” direction (equal to the rotation axis of the attitude matrix $g$), the two laws differ in the intensity of control action. Our feedback relies on the notion of group norm (as defined in equation (3)) and is proportional to this quantity. Instead the control laws proposed by Wen and Kreutz-Delgado are based on either the 2-norm of the unit quaternion or the 2-norm of the vector quaternion, and therefore exert an action proportional to either $\sin\|g\|$ or $2 \sin(\|g\|/2)$. 

IV. PD CONTROL ON SE(3)

Another common Lie group in robotic applications is SE(3). Unfortunately, since this Lie group is not compact, the results of the previous section cannot be applied directly. As before, we begin by studying the simplest first order case and we then couple proportional with derivative action for second order systems. Finally we apply our results to the case of mechanical manipulators and we then report some simulations.

IV.1. Proportional actions on SE(3) and first order systems

The geometric properties of the group SE(3) have received much attention in the recent control literature [3, 1] and a very complete treatment is contained in [15]. A well-known negative result is the following: no symmetric bilinear form on $\mathfrak{s}(3)$ can be both positive-definite and Ad-invariant. There is therefore an algebraic obstruction to the procedure we have followed for the SO(3) case.

Recall the design procedure: we need a positive-definite bilinear form (hence an inner product) to construct a Lyapunov function $W$, and we need the Ad-invariance of this form to compute the time derivative of $W$ (Theorem 6). Unfortunately the only bilinear forms on $\mathfrak{s}(3)$ are the following. Let $V_i = (\omega_i, v_i)$ for $i = 1, 2$ and consider

\[ \Box \text{a linear combination of Klein and Killing form:} \]

the most generic Ad-invariant form on $\mathfrak{s}(3)$ looks like

\[ \langle V_1, V_2 \rangle_{\text{Ad-inv}} = \alpha(\omega_1, \omega_2) + \beta(\omega_1, v_2) + (v_1, v_2), \]

\[ \Box \text{the standard inner product on} \mathfrak{s}(3) = \mathbb{R}^6; \text{discard the Lie algebra structure of} \mathfrak{s}(3) \text{ and write} \]

\[ \langle V_1, V_2 \rangle_{\mathbb{R}^6} = (\omega_1, \omega_2) + (v_1, v_2). \]  

(8)

Hence we are left with two possible design choices: as proportional action we can insist on the logarithm function (this corresponds to giving up the positive-definiteness), or (giving up the Ad-invariance) we can still regard SE(3) as a metric space with respect to the inner product (8) and compute the correct proportional action within this new framework.\footnote{Given an inner product on $g$, we can extend it to the whole TG by either left or right translation: we end up therefore with a metric structure on G. We refer to [6] for a detailed treatment of this standard construction.}

The two procedures are described in Figure 1; the following lemmas formalize this previous discussion.

Lemma 9. Consider the SE(3) system $\dot{y} = gV^b$ and let $k_p > 0$. Then the control law

\[ V^b = -k_p \log(g) \]

exponentially stabilizes the state $g$ at $I$ with time constant $k_p$ from any initial condition $g(0) = (R(0), p(0))$ such that $\text{tr}(R(0)) \neq -1$.

The second approach is based on the decomposition of the control system on SE(3) into a control system on $SO(3) \times \mathbb{R}^3$. Recall the notation introduced in Section 1: $g = (R, p)$, $V^* = (\omega^*, v^*)$, $V^b = (\omega^b, v^b)$. The original systems $\dot{g} = gV^b$ and $\dot{g} = V^by$ reduce to

\[ \begin{cases} \dot{R} = R\omega^b, \\ \dot{p} = Rv^b \end{cases}, \quad \text{and} \quad \begin{cases} \dot{R} = \omega^* R, \\ \dot{p} = \omega^* \times p + v^* \end{cases}. \]

Indeed adopting the bilinear form (8) involves applying a proportional action along geodesics for both the subsystems in $SO(3)$ and $\mathbb{R}^3$ (see [8] for more details).

**Fig. 1:** Proportional actions on SE(2): on the left the logarithmic function, on the right double-geodesics for $SO(2) \times \mathbb{R}^2$. Each point $g \in SE(2)$ is depicted as a frame on the plane.

**Lemma 10.** Consider the SE(3) system $\dot{g} = gV^b$ and let $K_w, K_v$ be positive definite symmetric gains. Then the control law

\[ \begin{cases} \omega^b = -K_w \log_{SO(3)}(R) \\ v^b = -R^T K_v p \end{cases}, \]

exponentially stabilizes the state $g$ at $I$, from any initial condition $g(0) = (R(0), p(0))$ such that $\text{tr}(R(0)) \neq -1$.

Similar versions of the two lemmas can be easily written for the right invariant case ($\dot{g} = V^g g$). Because of the basic Lie group identity $Ad_\omega \log(g) \equiv \log(g)$, it is easy to show that left and right invariant systems behave the same way under the logarithmic control law. This is not true for the double-geodesic law, as we shall see also in the simulations (again, see [8] for more details).

IV.2. Second order systems

We now apply these proportional strategies, coupled with a derivative term, to second order, fully actuated systems on SE(3). Consider the left invariant second order system on SE(3):

\[ \begin{cases} \dot{R} = R\omega^b, \\ \dot{p} = Rv^b \end{cases}, \quad \begin{cases} \dot{\omega} = gV^b \end{cases}, \quad \begin{cases} \dot{V} = f(g, V^b) + U, \end{cases} \]

where $f(g, V^b), U \in \mathfrak{s}(3)$ are internal drift and control input. The previous discussion leads to the two theorems:

**Theorem 11.** (Regulation via the logarithm function)

Consider the system in equation (10) and let $k_p > 0$ and $K_d = K_d > 0$. Then the control law

\[ U(g, V^b) = -f(g, V^b) - k_p \log(g) - K_d V^b, \]

locally exponentially stabilizes the state $g$ at $I \in SE(3)$.

**Theorem 12.** (Regulation via the double-geodesic law)

Consider the system in equation (10) and let $K_w, K_v$ and $K_d$ be the positive definite gains. Then the control law

\[ U(g, V^b) = -f(g, V^b) - K_w \log_{SO(3)}(R) \]

\[ \frac{R}{R^T K_v p} - K_d V^b, \]

exponentially stabilizes the state $g$ at $I$ from any initial condition $g(0) = (R(0), p(0))$ with $\text{tr}(R(0)) \neq -1$ and for all $K_w$ and $\omega^b(0)$ such that

\[ \lambda_{\min}(K_w) > \frac{\|\omega^b(0)\|^2}{\pi^2 - \|R(0)\|^2_{SO(3)}}. \]
Remark 13. As usual we can extend to the right invariant case ($\dot{g} = V^b g$) all we have done for the left one. For both systems the logarithmic control law (in Theorem 11) is identical and it can be shown that the closed loop vector field is exactly the same. Instead, the double–geodesic law applied to a right system has the slightly different expression:

$$U(g, V^b) = -f(g, V^b) - \left[ \frac{K_w \log_{SO(3)}(R)}{K_v} \right] - K_d V^b.$$

IV.3. Example: workspace control of robotic manipulators and 6 DOF underwater vehicles. As in Subsection III.1, we apply our control laws to mechanical systems. After a change of coordinates and inputs, 6 degree of freedom robotic manipulators and underwater vehicles can be modeled as

$$\begin{aligned}
\dot{g} &= g V^b, \\
M(g) \dot{V}^b &= C(g, V^b) V^b + N(g, V^b) + U,
\end{aligned}$$

where $M(g)$ is the inertia matrix, $C(g, V^b)$ is the Coriolis matrix and $N(g, V^b)$ are some generic extra terms. The kinetic energy of this mechanical system, is computed with the positive definite form (8) (coupled with the left translational of the velocity $g V^b$). Hence, for this class of systems, we are naturally led to prefer the double–geodesic control law over the logarithmic one:

$$U(g, V^b) = -N(g, V^b) - k_p \left[ \frac{\log_{SO(3)}(R)}{R^T p} \right] - K_d V^b.$$

Exponential stability is proved through the Lyapunov function

$$W(g, V^b) = \frac{k}{2} \left( \| R \|_{SO(3)}^2 + \| p \|^2 \right) + \langle V^b, M(g) V^b \rangle_{\mathfrak{so}(3)}$$

$$+ \epsilon \left( \left[ \frac{\log_{SO(3)}(R)}{R^T p} \right], M(g) V^b \right)_{\mathfrak{so}(3)}.$$

IV.4. Example: position and attitude stabilization of planar rigid body. To compare the two classes of controllers presented above, we consider the problem of stabilizing a planar rigid body. Note that the subgroup of the planar motions $SE(2)$ contains still most of the complexity and richness of the full $SE(3)$ case.

We have simulated the control law described in Theorem 11 and 12 (logarithmic and double–geodesic laws for left invariant systems), and in Remark 13 (double–geodesic law for right systems). As foreseen from theoretical considerations, the logarithmic control law generates the same closed loop trajectories for both the right and the left invariant systems; we report therefore only the plot for the left case. The shape of the trajectories varies considerably depending on the size of the initial angle error and on the gain values: for all three feedbacks, left double–geodesic, logarithmic and right double–geodesic, we picked an initial rotational error equal to $\pi/2$ and we choose the proportional and derivative gains as $k_p = 1, k_d = 1$.

Looking at the plots in Figure 2 a few simple remarks can be made. First of all, the rotational part of the state behaves independently on the choice of control law: this agrees with the fact that our choices are equal in this part. Second, all the three control laws seem to converge at a very similar rate in both the rotational (of course) and translational part. Eventually the clearest difference regards the opposite handedness of the various control laws. Corresponding to a choice of left invariant control system (hence let us exclude now the right system case) the logarithmic and double–geodesic feedbacks will follow quite different paths even from a simple qualitative viewpoint.

V. Trajectory Tracking

We describe here a general approach to trajectory tracking problems for second order systems defined on Lie groups. In particular, by exploiting the group structure we attempt to reduce the tracking problem to a stabilization one for an appropriately defined error system.

V.1. Basic properties of dynamical systems on Lie groups. We characterize here the behavior of the composition of systems defined on the group $G$. Recall that, given $L, R \in G$ and $l, r \in g$, we call $l = LR$ a left control system and $\tilde{r} = Rr$ a right control system. Also, given two systems with state $p(t), q(t) \in G$, we call the inverse system the one
corresponding to the state \( p(t)^{-1} \) and the product system the one corresponding to the state \( p(t)q(t) \). By performing some chain rule differentiations, we have:

**Lemma 14.** (Time derivative of composed systems) With the notations just introduced it holds

\[
\dot{t}^{-1} = -L^{-1}, \quad \dot{t}^{-1} = -t^{-1} R,
\]

and

\[
(pq)^{-1} \frac{dpq}{dt} = \text{Ad}_{q^{-1}} (p^{-1} p + q^{-1} q) \quad \frac{dpq}{dt} (pq)^{-1} = pp^{-1} + \text{Ad}_p (qq^{-1}),
\]

where the adjoint map \( \text{Ad} \) is defined in Section II.

By means of these basic results, we are able to describe straightforwardly how, for instance, the product of two left control systems evolves in time: let \( \dot{l}_1 = l_1 L_1 \) and \( \dot{l}_2 = l_2 L_2 \), we have

\[
\frac{d}{dt} l_1 l_2 = l_1 l_2 (\text{Ad}_{l_2}^{-1} L_1 + L_2).
\]  

**Lemma 15.** (Derivative of adjoint map) Let \( U(t) \in \mathfrak{g} \), \( l = LL \) and \( l = Rl \), with \( l, r \in G \) and \( L, R \in \mathfrak{g} \). Then

\[
\frac{d}{dt} (\text{Ad}_q(t) U(t)) = \text{Ad}_q(t) \dot{U} + \text{Ad}_q [L,U],
\]

\[
\frac{d}{dt} (\text{Ad}_q(t) U(t)) = \text{Ad}_q(t) \dot{U} + [R, \text{Ad}_q(U)].
\]

Now, recalling equation (11), we can define \( l_{12} \triangleq l_1 l_2 \) and \( V_{12} = \text{Ad}_r^{-1} L_1 + L_2 \); Lemma 15 gives:

\[
\begin{cases}
    \dot{l}_{12} = l_{12} V_{12}, \\
    V_{12} = \text{Ad}_r^{-1} L_1 + [\text{Ad}_r^{-1} L_1, L_2] + L_2,
\end{cases}
\]

which shows how we can write in full generality the second order dynamics of the combination of Lie group systems.

**V.2. Extending regulators to trajectory trackers** Assume we are given a left invariant, second order control system on \( G \)

\[
\begin{align*}
\dot{g} &= gV, \\
V &= f(g, V) + U,
\end{align*}
\]  

and a control law \( U = Z(g, V) \) that makes the closed loop driftless system \( \dot{g} = gV \) locally asymptotically (exponentially) stable at the identity \( e_G = I \). We want to design a control law that makes the state \( g \) track a reference trajectory \( g_d \in G \) described by \( g_d = V_d(g_d) \), \( V_d(t) \in \mathfrak{g} \). The following theorem gives a general solution:

**Theorem 16.** (Trajectory tracking) Consider the system in equation (12), the asymptotic (exponential) regulator law \( Z(g, V) \) and the desired trajectory \( g_d(t) \). Define the configuration error \( e = g^{-1} g_d \in G \) and the velocity error \( V_e = (\text{Ad}_{g^{-1}} V_d + V) \in \mathfrak{g} \). Then the control law

\[
U = U_\pi(g, V) + U_{\text{tr}}(g, V, V_d, \dot{V}_d) + Z(e, V_e)
\]

where

\[
\begin{align*}
U_\pi(g, V) &= -f(g, V) \\
U_{\text{tr}}(g, V, V_d, \dot{V}_d) &= \text{Ad}_{g^{-1}}(\dot{V}_d) + [\text{Ad}_{g^{-1}}(V_d), V].
\end{align*}
\]

makes the state error \( e \) locally asymptotically (exponentially) stable approach the identity \( I \in G \).

A few remarks: first, for the \( SO(3) \) case we can simplify the tracking law by defining \( U_{\text{tr}} = A_{g^{-1}} V_d \); the control law (6) would still ensure exponential stability thanks to the properties discussed in Subsection II.3. Second, in stating Theorem 16, we assumed our control system to be a left system and the desired trajectory system to be a right one. These two choices are suited to the kind of mechanical systems we are interested in, such as example satellite reorienting and robotic manipulation. However, similar versions of the theorem can be stated using right control systems and left desired trajectories as well as other possibilities.

**V.3. Choices of error function on \( SE(3) \)** So far our choice of state error has been \( e \triangleq g^{-1} g_d \). Decomposing this error in its rotational and translational components, we have:

\[
e = \begin{bmatrix}
R_d^T R & R_d^T (p - p_d) \\
0 & 1
\end{bmatrix}.
\]  

(13)

Also an equivalent approach would be considering \( g^{-1} g_d \equiv e^{-1} \). As described in Section IV controlling \( e \) or \( e^{-1} \) is the same control problem.

This choice of \( e = g^{-1} g \) has the strong physical interpretation that, if \( g \) represents the body frame and \( g_d \) the desired frame, then \( e \) is the relative \( g \) to \( g_d \) frame. If we discard this physical reasoning, other choices of configuration error are possible. In particular the following two appear appealing:

1. We define the **reciprocal error** as

\[
e_{\text{reciprocal}} \triangleq g_d^{-1} = \begin{bmatrix}
R_d^T & p - R_d^T \cdot p_d \\
0 & 1
\end{bmatrix}
\]

This definition has the drawback that even for \( p = p_d \), i.e. for overlapping frames, \( e_{\text{reciprocal}} \) sees some translational error if \( R \neq R_d \). This will reflect in a control law with non-zero translational input, which is undoubtedly undesired.

2. In keeping with the notion in [1], we define the **hybrid error** as

\[
e_{\text{hybrid}} \triangleq \begin{bmatrix}
R_d^T & p - p_d \\
0 & 1
\end{bmatrix}
\]

Even though this definition would seem rather natural, the Lie group structure of the original problem is not taken into account. It happens in particular that the value of the hybrid error between body and desired frame does depend on the arbitrary choice of inertial frame. To see this, consider a left translation \( g = (R_0, p_0) \): \( g \mapsto g' = g_0 g \) and \( g_0 \mapsto g_d = g_0 g_d \). Then it is easily seen

\[
e_{\text{hybrid}}' = \begin{bmatrix}
R_d^T & R_0 (p - p_d) \\
0 & 1
\end{bmatrix}
\]

Eventually note that the drawbacks just described affect also the inverse definitions \( e_{\text{reciprocal}}^{-1} \) and \( e_{\text{hybrid}}^{-1} \). By and large, we therefore prefer the standard choice (13), which is indeed adopted in Theorem 16.
VI. DISCUSSION

In this paper we have generalized proportional derivative control laws for systems in \( \mathbb{R}^n \) to systems on matrix Lie groups. In the compact case (e.g. \( SO(3) \)) we make use of the natural metric structure of the configuration space and give completely general results. Similarities with existing control laws by Wen and Kreutz [13] are discussed. We also design generalized PD control laws for \( SE(3) \), where no natural metric structure is present: different possible choices depend on whether \( SE(3) \) is treated as a direct or semi-direct product of \( SO(3) \) and \( \mathbb{R}^3 \) for the purposes of controller synthesis. We then show an additional advantage to using the group structure by extending controllers designed for stabilization to controllers for trajectory tracking. The group operation naturally defines a notion of error function with the same dynamics as the original system; as in the the \( \mathbb{R}^n \) case, we track the desired trajectory by stabilizing the error to zero. Also note that most of the results stated for the Euclidean group \( SE(3) \), have actually a much wider scope and hold for generic Lie groups.

Several directions for future research have not been explored in this paper. In particular control problems related to underactuated mechanical systems are of great importance. Indeed more insight into the underactuated case is gained with the methods we have illustrated. For example the group error function still seems a meaningful notion in trajectory tracking problems and for the point stabilization problem exponential coordinates appear a convenient group parametrization by providing homogeneous approximations to the original vector fields (see [16] and [8]). Also with respect to exponential coordinates the linearization of a Lie group system along a desired trajectory has a simplified form and the application of modern linear techniques appears straightforward.

Ultimately we plan to focus on mechanical systems with symmetries, where the use of geometric techniques allows the system dynamics to be split into a set of reduced dynamics and a principal connection which describes the reconstruction process (for a discussion see [17]). In this setting, the dynamics of the system have the form

\[
g = g(-A(x)\dot{x} + \Omega^{-1}(x)A\Omega\mu) + M(x)\dot{x} = N(x, \dot{x}) + u
\]

where \( x \in M \) describes the base manifold (shape space), \( g \in G \) gives the fiber coordinates, and the remaining notation is as described in [17]. We retrieve the equations considered here when \( A(x) = -I \), \( \mu = 0 \), and \( \dim(M) = n \).

REFERENCES


